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SHUTTLE CRITICAL ITEMS LIST - ORBITER

المسجورين

SUBSYSTEM : EWAI/MPS

FMEA NO 05-7J -2002 -1 REV:03/17/89

:SEE TABLE 05-7J-2002-1 LOCATION <u>P/N RI</u>

OUANTITY

CRIT. FUNC: CRIT. HDW: 1

1. NB6GE24-61XXXX 2. NBS6GEB-3XX

2 1

VEHICLE 102 103 104 105 EFFECTIVITY Х X X х

PHASE(S): PL X LO X CO DO LS

REDUNDANCY SCREEN: A- N/A B- N/A C- N/A

APPROVED BY (NASA):

PREPARED BY:

DES J BROWN REL

KIMURA COURSEN

APPROVED DES 16 m CZ 44

REL 16 QΕ

ENGI SSM & 4 Herrsten 4/19/89 EWAL REL-PULZ en worden your SSX REL 1 Bane 1 4/1-134

ITEM:

QΕ

CONNECTOR, PLUG, [(1.) 61 #20 CONTACTS, (2.) 3 #20 CONTACTS] - LOZ OVERBOARD BLEED VALVE CIRCUIT.

FUNCTION:

PROVIDES MATE/DEMATE CAPABILITY FOR WIRING WHICH CONTAINS CIRCUITS FOR ENERGIZING THE LO2 OVERBOARD BLEED VALVE CLOSE SOLENOID. FOR PLUG AND SHORTED PIN-TO-PIN PIN/SOCKET REFERENCE DESIGNATORS, SEE TABLE 05-7J-2002-1.

FAILURE MODE:

PIN-TO-PIN SHORT (GROUND)

CAUSE(S):

PIECE PART FAILURE, CONTAMINATION, VIBRATION, MECHANICAL SHOCK. PROCESSING ANOMALY, THERMAL STRESS

EFFECT(S) ON:

- (A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE (E) FUNCTIONAL CRITICALITY EFFECT:
- (A) LOSS OF CAPABILITY TO ENERGIZE THE LOS OVERBOARD BLEED VALVE CLOSE SOLENOID CIRCUIT DUE TO A CRITICAL PIN IN THE CONNECTOR SHORTING TO A GROUND RETURN PIN.
- (B) INABILITY TO CLOSE THE LOS OVERBOARD BLEED VALVE.

SHUTTLE CRITICAL ITEMS LIST - ORBITER

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SUBSYSTEM : EW&I/HPS

FMEA NO 05-7J -2002 -1 REV:03/17/85

(C.D) POSSIBLE LOSS OF CREW/VEHICLE WHERE A CONTINUED BLEED FLOW RESULT IN LOSS OF LOZ OVERBOARD WITH FAILURE OF BLEED DISCONNECT (PDID) CLOSE. BLEED DISCONNECT IS NOT CERTIFIED FOR CLOSURE UNDER FLOW CONDITIONS AND CANNOT BE CONSIDERED A REDUNDANT INHIBIT AGAINST OMERSOAFT FLOW. POSSIBLE RUPTURE OF DISCONNECT HOUSING AND/OR DOWNSTPEAD BLEED SYSTEM DUE TO WATER HAMMER. RESULTS IN LOSS OF APPROXIMATELY 1000 LBS OF PROPELLANT WHICH IS INSUFFICIENT TO CAUSE PREMATURE SSME SHUTDOWN.

IF THE LO2 BLEED VALVE FAILS TO CLOSE BEFORE T-0. THE LI2 BLEET DISCONNECT WOULD BE CLOSING WITH AN OXYGEN FLOW OF 4.1 LBS/SEC. THIRT; TWO PERCENT OF THIS FLOW WILL BE VAPOR. THE LO2 BLEED DISCONNECT IS NOT CERTIFIED FOR CLOSURE UNDER FLOW. HOWEVER, THE CLOSURE IS AT THE ACCELERATION RATE (T-0 UMBILICAL SEPARATION RATE) WHICH LIMITS THE IMPACT ENERGY ON THE VESPEL SEAL TO A LEVEL WHICH IS BELOW THE LIZ VESPEL IGNITION LEVEL (NOT PREVIOUSLY TESTED WITH THIS CONDITION). THE WATER HAMMER EFFECT GENERATED DURING THIS CLOSURE HAS BEEN CALCULATED TO BY APPROXIMATELY 60 PSIG. SYSTEM PROOF PRESSURE LEVEL IS 286 PSIG.

POSSIBLE AFT COMPARTMENT OVERPRESSURIZATION. FIRE/EXPLOSIVE HAZARD BOTH INTERIOR AND EXTERIOR TO THE VEHICLE. NO LCC EXISTS FOR VERIFICATION OF VALVE POSITION PRIOR TO T-O.

REFERENCE CIL 05-6J-2090-1

DISPOSITION & RATIONALE:

(A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE:

(A,B,C,D) DISPOSITION AND RATIONALE

REFER TO APPENDIX K:

ITEM NO. 1 - TYPE NB CONNECTOR, CIRCULAR, MINIATURE ITEM NO. 2 - TYPE NBS CONNECTOR, CIRCULAR, MINIATURE

(B) GROUND TURNAROUND TEST

V41ABO.200 - COMPLETE ELECTRICAL VERIFICATION (EVERY FLIGHT)
V41BIO.170 - LO2 OVERBOARD BLEED VALVE RESPONSE (EVERY FLIGHT)

(E) OPERATIONAL USE

NO CREW ACTION CAN BE TAKEN.

TABLE 05-7J-2002-1

PART NUMBER	PLUG REFERENCE DESIGNATOR	CONNECTOR LOCATION	PIM SOCKET DESIG
1. NB6GEZ4-61XXXX	\$5V77W1P320	AFT LCA-2	``-s
	56V77W1P320	AFT LCA-3	i∜≁s
2. NB56GE8-3XX	50V77W91P73J	ON LOZ OVBD BLEED VALVE CLOSE SOLENOID	A-C